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# Large impacts and their contribution to the water budget of the Early Moon

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## Abstract

The Earth/Moon system likely results from a giant impact between a Mars-size object and the proto-Earth. This high-energy impact leads to extreme conditions under which volatile elements would not normally be preserved in the protolunar disk. However, recent measurements of lunar samples highlight the presence of a non-negligible amount of water in the Moon's interior (from 1.2 to 74 ppm). The aim of the present work is to quantify the water contribution of the late accretion on the early Moon. Here, we use a 2D axisymmetric model with the hydrocode iSALE-Dellen to study the fate of a large impactor on a target body similar to the early Moon with a crust, a magma ocean, and a mantle. For this purpose, we compute different models to monitor the depth to which the impacted material is buried at the end of the impact event and the degree of devolatilisation of the impactor. Three parameters are explored: the crustal thickness (ranging from 10 to 80 km), the impactor radius (ranging from 25 to 200 km) and the impactor velocity (ranging from 1 to 4 times the target escape velocity). Our models show that impactors with a radius greater than 25 km impacting a partially molten lunar body with a crust thinner than 40 km could significantly contribute to the water content of the lunar mantle even for impact velocities close to the lunar escape velocity. For impact velocities greater than 3 times the target escape velocity, the impactor material is significantly molten

27 and its water content is devolatilised within the lunar atmosphere. Depending  
28 on the water content of the impactor material and the ability of the lunar magma  
29 ocean to maintain chemical heterogeneities, the late lunar accretion following  
30 the Moon-forming giant impact could explain the differences in water content  
31 among the lunar samples.

32 *Keywords:* early moon, impact cratering, numerical modeling, LMO, late  
33 accretion

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## 34 1. Introduction

35 Several tens of millions of years after the beginning of its accretion, the  
36 proto-Earth experienced a giant impact that led to the formation of the Earth-  
37 Moon system (*Hartmann and Davis, 1975; Woo et al., 2022*). This collision  
38 with a proto-planet named Theia, whose radius was between 2000 and 4000  
39 km, occurred when the core/mantle separation of the early Earth was almost  
40 fully achieved (*Cameron and Ward, 1976; Canup, 2004; Ćuk and Stewart, 2012*).  
41 Such a catastrophic event would have generated a large disk of molten debris  
42 or a synestia (*Lock et al., 2018*) that re-accreted to form the Moon. Hence,  
43 after its accretion, the early Moon was probably significantly to fully molten  
44 (*Smith et al., 1970*). A thick magma ocean made of molten silicate material  
45 (*Warren, 1985*) surrounded a relatively small metallic core compared to other  
46 terrestrial planets (*Wieczorek, 2009*). After rapid cooling of its surface, solidifi-  
47 cation occurred (1) from the bottom of the lunar magma ocean (LMO) forming  
48 a thick layer of olivine followed by the solidification of pyroxene material (*Char-  
49 lier et al., 2018*) and (2) at the surface of the Moon where a thin unstable crust  
50 formed in contact with the colder atmosphere (*Monteux et al., 2016a*). Towards  
51 the end of the magma ocean stage, at 75-80% crystallisation (*Snyder et al.,  
52 1992; Elkins-Tanton et al., 2011*), a thick anorthositic crust had formed on the  
53 surface of the Moon (*Wood et al., 1970*). It left a molten reservoir between the  
54 crust and the solidified deep lunar mantle (<1000 km-thick). The early Moon  
55 solidified over a timescale ranging from 10 Myrs (*Elkins-Tanton et al., 2011*)  
56 to  $\approx 200$  Myrs after its formation (*Warren, 1985; Maurice et al., 2020; Michaut  
57 and Neufeld, 2022*).

58

59 After the giant impact, the accretion of material continues and large to gi-  
60 ant impacts occurred with consequences still visible from lunar observations

61 (e.g. *Zhu et al.* (2019a)). Thirty-two impact basins formed before 3.8 Gyr by  
62 impactors with diameters of more than 50 km are currently preserved on the  
63 surface of the Moon (*Garrick-Bethell and Zuber*, 2009; *Kamata et al.*, 2015;  
64 *Morbidelli et al.*, 2018). The South Pole-Aitken (SPA) basin (with a diameter  
65 of about 2500 km) is the largest preserved impact basin and one of the oldest  
66 preserved structures observed on the Moon (*Garrick-Bethell and Zuber*, 2009;  
67 *Evans et al.*, 2018; *James et al.*, 2019). This crater could be the consequence of  
68 a large impact with an impactor of  $\approx 160$  km in diameter (*Potter et al.*, 2012;  
69 *Trowbridge et al.*, 2020; *Miljković et al.*, 2021). However larger impact events  
70 likely occurred throughout the lunar accretion and the Procellarum basin might  
71 be a remnant of such a giant impact (*Nakamura et al.*, 2012). The origin of  
72 the Procellarum basin, which is even larger (with a diameter of about 3200 km)  
73 and older is still under debate (*Andrews-Hanna et al.*, 2014; *Zhu et al.*, 2019b)  
74 . *Zhu et al.* (2019a) estimated that more than 200 basin-forming impacts (with  
75 crater diameter  $> 300$  km) occurred between the giant impact event and the end  
76 of crystallisation of the magma ocean. Large impacts likely contributed to the  
77 final chemical and thermal evolution of the Moon (*Bottke et al.*, 2010). These  
78 large impacts are expected to have occurred in a context where the Moon was  
79 still significantly molten (see e.g. *Miljković et al.* (2021)), favouring chemical  
80 exchanges between the impactor material and the magma ocean beneath the  
81 solid crust.

82

83 The water content within the lunar interior has been estimated from H mea-  
84 surements carried out on volatile-bearing phases (see references in *McCubbin*  
85 *et al.* (2023): lunar apatites, glass beads and mesostasis glasses, melt inclusions,  
86 and nominally anhydrous minerals. It is non-trivial to determine the lunar water  
87 content in the lunar interior from these measurements because several processes

88 such as spallation or degassing may have modified this content and/or might be  
89 representative of an evolved magmatic source that is chemically different from  
90 the bulk lunar mantle. Chemical analyses of the lunar rocks sampled from the  
91 Apollo mission initially indicated that the lunar rocks were depleted in moder-  
92 ately volatile elements (*Lunar Sample Preliminary Examination Team*, 1969).  
93 Moderately volatile elements are elements with condensation temperatures be-  
94 tween those of Mg-silicates and FeS (*Palme et al.*, 1988). *Borg et al.* (2022)  
95 highlighted that the absence of these moderately volatile elements could have  
96 been caused by the formation of the two bodies involved in the creation of the  
97 Earth-Moon system. Moreover, the Moon formation scenario favours extreme  
98 conditions (with a temperature increase of at least  $\sim 5000$  K) and, as a conse-  
99 quence, a depletion in volatile elements in the Moon’s interior (*Canup*, 2008).  
100 Due to high impact temperatures and outgassing of the LMO, the Moon’s inte-  
101 rior should have experienced an important depletion in volatile elements during  
102 its accretion (*Boyce et al.*, 2015). However, more recent chemical analyses of lu-  
103 nar samples have shown a variable water content from 1.2 to 74 ppm within the  
104 lunar mantle (*Saal et al.*, 2008; *Tartèse et al.*, 2013; *Füri et al.*, 2014; *Nakajima*  
105 *and Stevenson*, 2018; *Stephant et al.*, 2020; *Hu et al.*, 2021). The wide range in  
106 water content within lunar samples seems to represent a degree of heterogeneity  
107 within the lunar mantle in terms of water content (*Robinson and Taylor*, 2014;  
108 *Robinson et al.*, 2016; *McCubbin et al.*, 2023).

109

110 The solidification processes that occurred within the Moon, combined with  
111 the occurrence of large impacts while molten reservoirs were still present (*Perera*  
112 *et al.*, 2018; *Zhu et al.*, 2019a) may have favored the incorporation of hydrated  
113 objects deep within the lunar mantle. Large impacts likely contributed to the  
114 water budget of the Moon (*Bottke et al.*, 2010) and induced significant produc-

115 tion of ejectas, melt and vapor (*Pierazzo et al.*, 1997; *Luo et al.*, 2022). The  
116 contribution of the impactors to the lunar volatile content is strongly depen-  
117 dent on the impact velocity (*Svetsov and Shuvalov*, 2015). Tracking impactor  
118 material within impacted bodies is now possible with the improvement of nu-  
119 merical modeling. *Potter and Collins* (2013) focused on the likelihood of the  
120 impactor remaining solid as a function of its characteristics (porosity, shape,  
121 impact velocity and angle) and proposed possible scenarios in which this mate-  
122 rial would remain within the crater. Then, *Svetsov and Shuvalov* (2015) focused  
123 on low-velocity impacts to determine the water content and the mass fraction  
124 remaining at the bottom of the craters with the Eulerian method in the iSALE  
125 hydrocode. More recently, *Kendall and Melosh* (2016) investigated the iron core  
126 stretching of a large impactor in a terrestrial magma ocean using tracers with  
127 the iSALE-3D hydrocode model. Finally, *Marchi et al.* (2018) used a hydrocode  
128 with the SPH method to track the iron core of a large impactor with a more  
129 realistic  $45^\circ$  angle.

130

131 In the present study, we monitor the fate of an impactor and its associated  
132 depth distribution in a magma ocean beneath a solid crust. Our aim is to con-  
133 strain the water contribution of large impacts at the time when the deep Moon  
134 was still significantly molten. For that we use the hydrocode iSALE to investi-  
135 gate the ability of an impactor to penetrate deep into the interior of the Moon  
136 as a function of three parameters: the lunar crustal thickness, the impactor ra-  
137 dius and the impact velocity. Our manuscript is organised as follows: In section  
138 2 we describe our physical and numerical models. We illustrate our modelling  
139 approach with a reference model in the section 3. In the section 4 we monitor  
140 the fate of the impactor using a parametric study. The section 5 discusses the  
141 evolution of the pressure and temperature of the impacted material as a func-

142 tion of the three parameters described above. In the section 6 we discuss the  
143 ability of large impacts to contribute to the water budget of the LMO and the  
144 limits of our model.

145

## 146 **2. Impact modelling: Methods**

147 During the first few Myrs after the giant impact that led to the Moon's  
148 formation, more than 200 impact basins (with diameters of more than 300 km)  
149 were likely formed (*Zhu et al.*, 2019a). These impacts occurred while molten  
150 reservoirs were likely still present within the deep Moon. In our study, we per-  
151 formed numerical hydrocode simulations to model a large impact event on a  
152 Moon-sized body with a magma ocean. For that, we used iSALE-2D shock  
153 physics code (*Wünnemann et al.*, 2006), which is based on the SALE hydrocode  
154 solution algorithm (*Amsden et al.*, 1980; *Melosh et al.*, 1992; *Ivanov et al.*, 1997).  
155 The code iSALE-2D is a multi-rheology, multi-material model that enables to  
156 model large impacts on a planetary scale (*Monteux et al.*, 2014; *Bray et al.*, 2014;  
157 *Monteux and Arkani-Hamed*, 2019; *Trowbridge et al.*, 2020). iSALE simultane-  
158 ously solves the equations of conservation of mass, energy and momentum with  
159 constitutive relations for material properties. We used the most recent version  
160 which is iSALE-Dellen (*Collins et al.*, 2016).

161

162 During a large impact on a rocky body, three stages occur: contact and  
163 compression stage, excavation stage and the modification stage (*Melosh*, 1989;  
164 *Collins et al.*, 2012). The excavation phase leads to the opening of a crater  
165 while the modification stage alters its structure over a longer timescale to its  
166 final state. The presence of a molten layer within the interior of the impacted  
167 target may influence the deformation processes during the impact. *Monteux*



168 *et al.* (2016b) showed that the presence of a thick water ocean within Enceladus  
169 was reducing the impact-induced deflection of the solid core surface beneath the  
170 ocean. More recently, *Miljković et al.* (2021) showed the influence of a thin melt  
171 layer on the basin-forming process, where magma flow dominates the crater col-  
172 lapse and changes its final morphology. In the following sections, we describe  
173 our modelling approach to a large impact on a partially molten Moon.

174

### 175 *2.1. Characteristics of the impact*

176 After the Moon forming impact, large impacts continued to deliver material  
177 to the Moon. The distribution of impact velocities on the Moon is similar to  
178 that for the Earth, although they are shifted to lower values because the Moon  
179 has a lower gravity. If coming from the asteroid orbit, the impact velocity dis-  
180 tribution has a maximum at about  $12 \text{ km s}^{-1}$  (*Bottke et al.*, 2002). This peak  
181 in velocity distribution is a function of the impactor size and ranges from 10 to  
182  $15 \text{ km s}^{-1}$  (*Marchi et al.*, 2009). A non negligible fraction of the lunar crater oc-  
183 curred with a velocity lower than  $10 \text{ km s}^{-1}$  (*Ito and Malhotra*, 2010; *Le Feuvre*  
184 *and Wieczorek*, 2011), the lower bound of the impact velocity being the lunar  
185 escape velocity ( $= 2.38 \text{ km s}^{-1}$ ). In our models we use impact velocities  $v_{imp}$   
186 ranging between 1 and 4 times the escape velocity  $v_e$  of our modelled planet  
187 (corresponding to 2.33 and  $9.32 \text{ km s}^{-1}$  respectively). The upper bound of im-  
188 pact velocity is chosen to limit numerical instabilities and corresponds to the  
189 typical velocity on the Moon (*Marchi et al.*, 2012).

190

191 Impact angles ( $\theta$ ) potentially range between  $0^\circ$  and  $90^\circ$  with respect to the  
192 tangent of the impacted planet's surface ( $90^\circ$  being a vertical impact). Accre-  
193 tionary models that have taken into account moderately oblique impacts do not  
194 show a significant influence of impact angle on the final state of an accreted body

195 (*Agnor and Asphaug, 2004; Kokubo and Genda, 2010*). However, decreasing the  
196 impact angle from  $90^\circ$  to  $45^\circ$  (the most probable impact angle (*Shoemaker,*  
197 *1962*)) leads to a reduction in shock pressure by a factor of  $\sim 0.71$  (see *Pierazzo*  
198 *and Melosh (2000a)*). For extremely low impact angles (with  $\theta \leq 30^\circ$ ), the im-  
199 pactor will not penetrate into the target (e.g. *Elbeshausen et al. (2009)*). In our  
200 models and as first step, we consider only head-on collisions (impact angle of  $90^\circ$   
201 to the target tangent plane). The role of impact angle is left to future studies.  
202 Hence, we used 2D cylindrical geometry models, less costly in computation time  
203 than a 3D model.

## 204 *2.2. Characteristics of the impactor*

205 The ability of an impactor to contribute to the water content of the impacted  
206 Moon is governed by its composition and its origin. The water could have been  
207 delivered to the Moon by comets and/or asteroids (*Bruck Syal and Schultz,*  
208 *2015; Barnes et al., 2016*). According to *Nesvorný et al. (2023)*, the contribu-  
209 tion of the comets is not significant during the early bombardment of the Moon.  
210 Recent studies have shown that the early impactors of terrestrial planets are  
211 likely to be leftover planetesimals formed in the terrestrial planet region (*Mor-*  
212 *bidelli et al., 2018; Zhu et al., 2021; Nesvorný et al., 2023*). In this scenario, the  
213 impactors would likely be differentiated and the water contribution of the late  
214 accretion could be relatively poor. If we consider aubrites, that are achondrites  
215 produced by igneous differentiation of the enstatite chondrite, they have water  
216 contents three orders less than those of ECs (*Lorenz et al., 2021*). Other studies,  
217 however, argue that hydrogen was delivered to the Earth/moon system by ac-  
218 cretion of chondritic material (*Tartèse and Anand, 2013; Saal et al., 2013; Füri*  
219 *et al., 2014*). More recent models show that the accretion of terrestrial plan-  
220 ets is a heterogeneous process, whereby predominant accretion of volatile-poor  
221 planetesimals was followed by a second volatile-rich stage from undifferentiated

222 meteorites (*Liu et al.*, 2023). As a consequence and for simplicity, we assume  
223 that the potential water contribution from the impactors range between water-  
224 poor (enstatite chondrites) to volatile-rich (CI chondrites). However, depending  
225 on the type of chondrite envisioned, large differences in the water content may  
226 arise among the impactors. Nearly 500 planetesimals with a radius larger than  
227 10 km have impacted the Moon since its formation (*Nesvorný et al.*, 2023). The  
228 presence of large impact craters such as SPA or Procellarum suggest that the  
229 range of impactor radii may exceed 100 km in diameters (e.g. *Melosh et al.*  
230 (2017); *Zhang et al.* (2022); *Jones et al.* (2022)). In our study, we consider that  
231 the impactor radius ( $R_{imp}$ ) ranges between 25 and 200 km.

232

233 Depending on its evolution, a large metallic core is also likely to be present  
234 within the impactor. However, unless they formed rapidly, even large impactors  
235 might have been too small to have experienced metal/silicate separation (*Ri-*  
236 *card et al.*, 2017). If the impactors are leftover planetesimals formed at the  
237 early Solar System as suggested by *Zhu et al.* (2021) or *Nesvorný et al.* (2023),  
238 the impactors with a diameter larger than 10 km could be differentiated. For  
239 simplicity, and as a first step towards more complex models, the impactor is  
240 assumed to be a spherical body with a uniform solid composition. The presence  
241 of a metallic core within the impactor would increase its mass and as a conse-  
242 quence its kinetic energy. Hence, considering a homogeneous impactor in our  
243 models is a conservative assumption.

244

### 245 2.3. Characteristics of the target

246 According to the giant impact scenario, the Moon’s mantle is thought to be  
247 derived from the silicate Earth’s mantle (*Canup*, 2012; *Ćuk and Stewart*, 2012;  
248 *Hosono et al.*, 2019). After its re-accretion, the lunar mantle remained signif-

249 icantly molten with an estimated magma ocean depth of 400 km to 1200 km  
250 (*Snyder et al.*, 1992; *Charlier et al.*, 2018; *Shearer*, 2006). During the solidifi-  
251 cation of a potentially fully molten lunar mantle, the first mineral to crystallise  
252 would have been olivine. The crystallisation of the olivine likely began from  
253 the bottom of the magma ocean and formed a 300-500 km-thick solid layer.  
254 Upon cooling, orthopyroxene and then clinopyroxene phases would have solidi-  
255 fied (*Solomatov*, 2007; *Elkins-Tanton et al.*, 2011; *Charlier et al.*, 2018; *Johnson*  
256 *et al.*, 2021; *Kraettli et al.*, 2022). When the magma ocean was  $\approx 75\text{-}80\%$  so-  
257 lidified, fractional crystallisation of the residual melt resulted in the formation  
258 of the anorthositic lunar crust by floatation of the plagioclase crystals (*Snyder*  
259 *et al.*, 1992; *Elkins-Tanton et al.*, 2011). During this last crystallization phase,  
260 a thin crust formed over the remaining molten material. In the classical models  
261 of magma ocean fractional crystallization, the thickness of the magma ocean  
262 when the anorthite lunar crust starts to form ranges between 110 and 200 km  
263 (*Maurice et al.*, 2020). Assuming that the convection is vigorous enough to  
264 maintain a slushy material with a crystal fraction of up to 50%, the thickness of  
265 the slush (that should behave as a liquid) may encompass the full thickness of  
266 the lunar mantle when the crust starts to form (*Michaut and Neufeld*, 2022). In  
267 our study, we consider impacts occurring early during the lunar magma ocean  
268 cooling and arbitrarily fix the bottom of the magma ocean to 300 km. Hence the  
269 magma layer in our models is much thicker compared to 10-50 km as used in  
270 *Miljković et al.* (2021).

271

272 Hence, for 150 to 200 million years after the Moon forming impact, a magma  
273 ocean was encompassed between a thick solid mantle and a crust with a thickness  
274 increasing through the lunar cooling process (*Maurice et al.*, 2020). Currently,  
275 the entire silicate material of the Moon is supposed to be solid and its crust

276 is on average between 34 and 43 km thick with maximum value up to 60 km  
277 (*Wieczorek et al.*, 2013). In our study, we modelled the collision between an  
278 impactor made of silicate material and a Moon-size target (with radius  $R = 1740$   
279 km) composed of three homogeneous layers:

- 280 1. a solid dunitic mantle with a constant thickness 1440 km-thick,
- 281 2. an intermediate dunitic magma ocean with a thickness (220 to 290 km)  
282 that depends on the chosen thickness for the lunar crust,
- 283 3. a thin solid crust. To identify the influence of the crustal thickness on the  
284 penetration ability, we consider a range of crustal thickness values with an  
285 upper bound value larger than the lunar crustal thickness. We consider a  
286 crustal thickness ranging between 10 and 80 km.

287 In our models, we do not consider the Moon's core is to be affected by the  
288 impacts. The core size of the Moon is expected to be small compared to its  
289 mantle thickness (with a radius  $\approx 380$  km according to *Garcia et al.* (2011)). In  
290 our models, we do not include the presence of a dense metallic iron core within  
291 the impacted Moon. This results in a surface gravity and, as a consequence,  
292 an escape velocity that are slightly lower on our modelled lunar target ( $= 2.33$   
293  $\text{km s}^{-1}$ ) than on the Moon ( $= 2.38 \text{ km s}^{-1}$ ). This difference in surface gravity  
294 is small enough to not affect neither the cratering formation processes nor our  
295 results and interpretations.

296

#### 297 *2.4. Properties of the involved material*

298 The thermodynamic parameters (pressure, volume, temperature, internal  
299 energy) of the materials from our models are characterised by equations of  
300 state (EOS) for each material (crust, magma ocean, mantle and impactor). To  
301 characterise the material behaviour during the impact, we use two chemical rock

302 compositions with their corresponding EOS: granite for the crustal material and  
303 dunite for the impactor, the magma ocean and the mantle. The thermo-dynamic  
304 properties and equations of state for chondritic material are not available in the  
305 materials proposed by iSALE. Hence we use dunitic material for the impactor  
306 material (*Pierazzo and Melosh, 2000b; Zhu et al., 2019a*). We used the semi-  
307 analytical EOS (ANEOS) to model the thermodynamic behaviour of the dunitic  
308 material (*Benz et al., 1989; Thompson, 1990*) (see Tab. 1). The Moon's crust is  
309 anorthositic and was formed by plagioclase floatation (*Smith et al., 1970; Wood  
310 et al., 1970*). However, in iSALE Dellen no such material is available. Hence, we  
311 used the ANEOS model of granitic material whose density is similar to the den-  
312 sity of the anorthosite. This choice is relevant for modeling the thermodynamic  
313 response of the anorthositic crustal material during a large impact event (*Melosh  
314 et al., 2013; Johnson et al., 2018; Trowbridge et al., 2020; Miljković et al., 2021*).

315

316 During an impact, the deformation of intact rock occurs in different steps.  
317 When a constant pressure is applied to the rock, it first deforms elastically, pro-  
318 portionally to the stress without accumulating damage up to a certain threshold.  
319 Then it deforms plastically and the rock damage is accumulated permanently  
320 until fracturing occurs. At this point, the damage is maximal, the rock becomes  
321 fully fractured and the strength is controlled by rock friction (*Collins et al.,  
322 2004*). In this context, the material strength strongly influences the crater  
323 morphology as well as the post-shock temperature increase within the impact  
324 site. Hence considering a relevant strength model is crucial for modelling the  
325 deformation of both the impactor and the target and their temperature evolu-  
326 tions (*Wakita et al., 2019, 2022*). iSALE includes numerous strength models  
327 for solid and granular material, ranging from simple (Lundborg) to complex  
328 (Johnson-Cook). For solid material (crust, impactor and mantle) we used the

329 ROCK model, which is a combination of Lundborg’s equations with a damage  
330 parameter computed by the damage model (see Tab. 1). With this model, the  
331 deformation is sensitive to the state of the material which becomes weaker as  
332 the damage increases. For liquid behaviour, there are two models in iSALE:  
333 a Newtonian fluid model (with viscosity) and an inviscid fluid model (without  
334 viscosity). The magma ocean is a thick layer of molten rock with low viscosity  
335 in which there is vigorous convection (*Solomatov, 2007*). In such a reservoir, the  
336 magma viscosity may range between  $10^{-1}$  and  $10^2$  Pa s (*Solomatov, 2007*). Our  
337 simulations assume a fully molten inviscid magma ocean to reproduce purely  
338 hydrodynamic behaviour (*Kendall and Melosh, 2016*).

339

340 The strength model influences how the kinetic energy from the impactor is  
341 converted into heat within both the impactor and the impacted material. Dur-  
342 ing an impact, the shock compression is the main contributor to the impact  
343 induced heating. However, for low-velocity impacts (with  $v_{imp} \leq 10$  km s $^{-1}$ )  
344 both the compression and decompression have to be accounted for the calcu-  
345 lation of the impact-induced temperature increase (*Pierazzo et al., 1997; Zhu*  
346 *et al., 2017; Kurosawa and Genda, 2018; Manske et al., 2021*). *Kurosawa et al.*  
347 (2021) also illustrated that plastic deformation during low-speed collisions (with  
348  $v_{imp} \leq 10$  km s $^{-1}$ ) efficiently converts kinetic energy into internal energy within  
349 the shocked rocks. As a result, the degree of impact heating during low-speed  
350 collisions should be higher than if only the shock compression was considered.  
351 As a consequence, plastic work and decompression by structural uplift also con-  
352 tribute to melt production (*Manske et al., 2022*). A common approach to quan-  
353 tify impact induced melt production is to monitor the amount of material that  
354 experiences shock pressures larger than the critical shock pressure for melting  
355 (e.g. *Pierazzo et al. (1997)*). In our study, we estimate the temperature in-

crease by monitoring directly the temperature from the iSALE models from the beginning of the contact up to nearly 4000 s after the impact. This method is similar to the final temperature method and is relevant for low-velocity impacts (*Quintana et al.*, 2015; *Manske et al.*, 2022). We then compare the obtained temperature with the melting temperature of chondritic material for a given pressure to estimate the melt fraction. This method may suffer from artificial diffusion (*Artemieva*, 2007). However, *Manske et al.* (2022) have shown that, in the early stage of the simulations, this direct temperature method provides accurate results of the total melt production.

A damage model is used to describe how the strength transition between an intact and a fractured material occurs. Three damage models are available in the iSALE code. The simple model takes a damage parameter that is a linear function of the plastic strain. In the *Ivanov et al.* (1997) model, damage is also a function of plastic strain. However, the plastic strain when rupture is reached is an increasing function of the total pressure  $P$  (lithostatic pressure plus the pressure induced by the shock). This model allows for rapid brittle failure under tension at low pressure and semi-brittle failure at high pressure. In the damage model developed by *Collins et al.* (2004), the plastic deformation at failure is a function of the brittle-ductile and brittle-plastic transition pressures. Our study focuses on impact processes occurring at relatively low impact velocities and inducing relatively low impact pressures. In this context, the Hugoniot Elastic Limit (HEL) is not necessarily overcome and both the impacted material and the impactor can fail in tension (*Wiggins et al.*, 2019). To characterize the influence of the failure in tension at very low impact velocity, we have compared models including a Ivanov damage model with models explicitly accounting for tensile failure: the Collins damage model. In the Collins model, the tensile strength



383 is explicitly stated as  $Y_t = Y_{t0}(1 - D)$  with  $Y_{t0}$  the intact tensile strength  
384 and  $D$  the damage. Our results (not shown here) indicate that considering  
385 a Collins damage model neglecting tensile failure enhances the penetration of  
386 the impactor that penetrates 13% deeper than the Ivanov damage model at  
387 the end of the simulation. Including the tensile failure (with a tensile strength  
388 parameter of 10 MPa (*Trowbridge et al., 2020*)) in the Collins model decreases  
389 this depth difference from 13% to 8%. Hence, neglecting tensile failure leads to a  
390 slightly overestimation of the penetration of the projectile. For the lowest impact  
391 velocities, the difference in the penetration depth between the Ivanov damage  
392 model and the Collins damage model including tensile failure is only 4%. Our  
393 conclusion is that including the tensile strength increases the resistance of the  
394 impacted material and decreases the depth reached by the impactor's material.  
395 However, the influence of the tensile failure appears to be less important than  
396 the choice of the damage model itself on the penetration ability of the impactor.  
397 Since the parameters in the Collins damage model are poorly constrained in the  
398 context of primitive protoplanets, we choose to use the Ivanov model as our  
399 damage model (*Ivanov et al., 1997*).

400

401 *Ostrowski and Bryson (2019)* reported the average porosity for different  
402 types of chondritic material in the Solar System. The enstatite chondrites have  
403 the lowest porosity (between 2.1 and 3.7%) while the CI chondrites are com-  
404 posed mainly of hydrated phyllosilicate matrix and have a porosity of nearly  
405 35%. Porosity may play a important role in the efficiency of impact heat-  
406 ing (*Davison et al., 2010*) and on the dehydration processes (*Sekine et al.,*  
407 *2015*). iSALE allows to include the influence of porosity on impacts models  
408 (*Wünnemann et al., 2006*). However, only three materials are allowed when  
409 modelling large impacts using iSALE with *Planet mode*. In our models, we

410 already consider three materials: the crust, the magma ocean and the mantle.  
411 To avoid a fourth material, both the impactor and the mantle are made of the  
412 same material (dunite). We note here that using dunitic material (with a low  
413 porosity) as a proxy for CI chondritic material (with a high porosity) is a strong  
414 assumption since the dunite should be significantly denser than CI chondrites.  
415 Adding porosity to this dunitic material means lowering the densities of the  
416 impactor and the impacted mantle. Hence, the impacted mantle would become  
417 less dense than the overlying magma ocean resulting in gravitational instabili-  
418 ties and potential large scale overturns. *Davison et al.* (2010) showed that for  
419 high velocity impacts between planetesimals, the impact heating was increasing  
420 with porosity. However, according to their models, this effect is relatively small  
421 for porosities lower than 30% and for the range of impact velocities envisioned  
422 in our study. Therefore, we do not include the effect of porosity in our models.

423

424 We used a pre-impact temperature profile that is conductive in the crust and  
425 convective in magma ocean (Figure 1). The choice of gradient influences the  
426 depth at which the neutral buoyancy of the impactor material will be reached  
427 within the magma ocean. For each crustal thickness considered in our study, the  
428 temperature gradient meets the solidus of the dunitic material at the bottom of  
429 the magma ocean.

430

431 Large impact simulations are likely to favor weakening processes (e.g. low  
432 density weakening or acoustic fluidization) facilitating deep-seated uplift of the  
433 crater floor and collapse of the rim (*Collins*, 2014). These processes result in  
434 complex craters formation. Material weakening by thermal softening is espe-  
435 cially efficient for large-scale events (*Potter et al.*, 2012). This process strongly  
436 depends on the pre-shocked temperature profile that is still poorly constrained

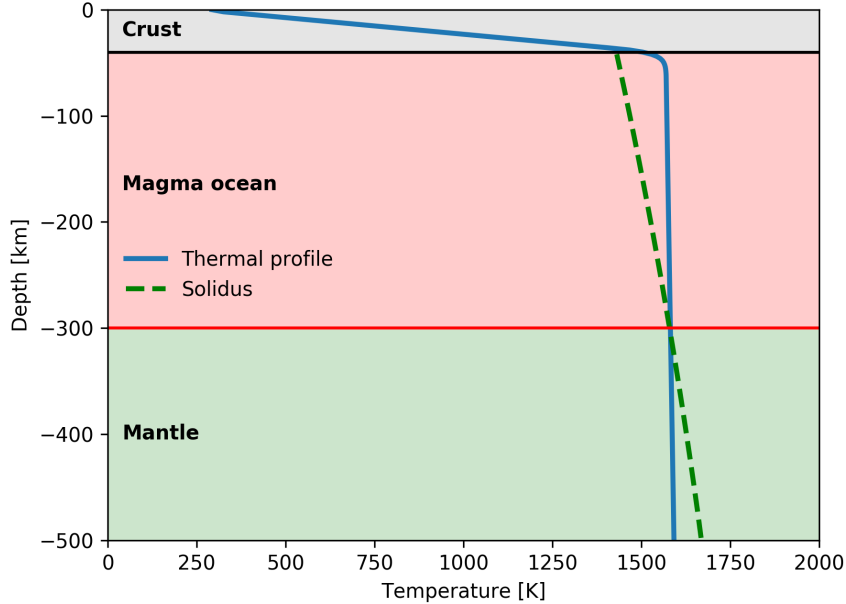


Figure 1: Initial temperature profile (blue curve) for the reference model considering a 40 km thick crust. The green curve represents the dunitic solidus from *Andrault et al. (2018)*.

437 during the early evolution of terrestrial planets. Exploring the influence of this  
 438 process would necessitate the exploration of a wide range of pre-impact tempera-  
 439 ture profiles that is beyond the scope of this study. In the context of a liquid  
 440 layer encompassed between two solid layers, this process might be of weak im-  
 441 portance (*Monteux et al., 2016b*). In our models, the acoustic fluidization is  
 442 likely to play a minor role on the shockwave propagation compared to that of the  
 443 strength. For simplicity and to reduce the number of free parameters, we chose  
 444 to neglect the low density weakening, the thermal softening and the acoustic  
 445 fluidization. The influence of the processes (that can be accounted with iSALE  
 446 models) favoring weakening and potentially affecting the impactor penetration  
 447 will be explored in a separated study.

### 448 3. Reference model

449 For a better illustration of the different processes occurring during a large  
450 impact, we first consider a reference model from which we highlight the main  
451 stages of the temperature and pressure evolutions. In this reference model, we  
452 consider a 40 km thick crust, an impactor radius of  $R_{imp} = 50$  km and an im-  
453 pact velocity  $v_{imp} = v_e = 2.33$  km s<sup>-1</sup>.

454

#### 455 3.1. Description

456 Figure 2 shows the time evolution of the compositional (left) and temper-  
457 ature (right) fields during a large impact for this reference case (see Table 2).  
458 The temperature field is saturated to illustrate the impact-induced temperature  
459 increase in the impactor during each stage. Figure 2 (Top cross-section) shows  
460 the initial configuration of the two bodies before the impact. The cross-section  
461 obtained at  $t = 315$  s shows the excavation stage leading to the formation of  
462 the transient crater on the Moon. In our study, we consider that the excava-  
463 tion stage ends when the transient crater reaches its maximum volume. The  
464 temperature increase ranges between 700 and 1700 K for the part of the im-  
465 pactor that is directly in contact with the Moon. The cross-section obtained at  
466  $t = 1200$  s illustrates the modification stage where the crater walls horizontally  
467 collapse followed by a vertical magma jet rising from the center of the crater.  
468 Vertical jets naturally form along the axis of symmetry and can be artifacts  
469 of the model setup. The heights of the jet obtained from our models do not  
470 show any exaggeration of this process (see Data availability Section). Similar  
471 processes are obtained from experimental analog models (*Landeau et al., 2021*).  
472 The corresponding inflow generates a supplementary pressure increase. During  
473 this process, 0.81% of tracers reach a pressure increase of more than 5 GPa. This  
474 pressure increase is smaller than that generated during the contact/compression

475 stage where 47% of tracers demonstrate a pressure increase of more than 5 GPa  
476 (see also Section 5). During the modification stage, the average subsequent  
477 supplementary temperature increase is  $\approx 120$  K in the impactor material. Our  
478 reference model is a low velocity impact, hence the peak temperature and pres-  
479 sure increases are moderate. Depending on the impact parameters, it is possible  
480 for the impactor material to penetrate the crust and reach the magma ocean.  
481 In this reference case, the end of the simulation is fixed at  $t = 4170$  s. At this  
482 time the impactor material has reached its final depth. A large crustal hole  
483 whose dimensions depend on the crustal thickness, impactor radius and impact  
484 velocity may remain at the lunar surface.

485

486 During the impact process, the impact energy dissipates within the impacted  
487 body as a function of time. As a consequence the subsequent deformation and  
488 movement of both the impactor and the impacted body decrease until an equi-  
489 librium is reached. To prevent the development of undesirable effects that are  
490 unrelated to the impact (e.g. numerical diffusion) and to save computation  
491 time, we included a simulation stop criterion based on the ratio between the  
492 average mass-weighted velocity of the impactor material obtained from the Eu-  
493 lerian method and the initial impact velocity. We stopped our simulations when  
494 this ratio fell below an arbitrary threshold value which we chose to fix at 1%.  
495 Figure 3 shows the ratio of the average of the velocity weighted to the mass of  
496 the impactor material over the initial impact velocity as a function of time for  
497 the reference model. This figure shows that the velocity of the impactor ma-  
498 terial decreases as a function of time during the early impact processes before  
499 oscillating during the late impact stages. These oscillations are due to the iso-  
500 static adjustment of the impactor in the magma ocean. In this reference model,  
501 we stopped our calculation  $t = 4170$  s after the beginning of the impact. At

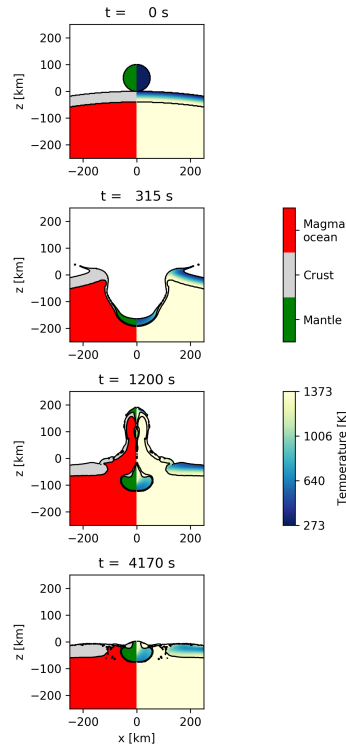


Figure 2: Evolution of the composition (left) and temperature (right) as a function of time. In our reference case ( $v_{imp} = v_e = 2.33 \text{ km s}^{-1}$  and  $\delta_{crust} = 40 \text{ km}$ ), the undifferentiated dunitic impactor ( $R_{imp} = 50 \text{ km}$ ) impacts a body with a 1740 km radius. The planet is composed of a shallow crust (gray), a magma ocean (red) and a mantle (green). The cross-section for  $t = 0 \text{ s}$  represents the conditions before the impact. The cross-section for  $t = 315 \text{ s}$  illustrates the beginning of the modification stage. The cross-section for  $t = 1200 \text{ s}$  shows the maximum peak height. Lastly, the cross-section for  $t = 4170 \text{ s}$  represents the state at the end of the simulation.

502 the end of our simulation, the impactor has penetrated the crust but remains  
 503 at shallow depth within the magma ocean below a highly fragmented crust.

504

### 505 3.2. Resolution test

506 iSALE simultaneously solves the conservation equations with constitutive re-  
 507 lations for material properties. These equations can be solved by simultaneously  
 508 applying the Eulerian (grid domain) and a Lagrangian (tracers) hybrid meth-  
 509 ods. The tracers are particularly relevant for following the impactor material

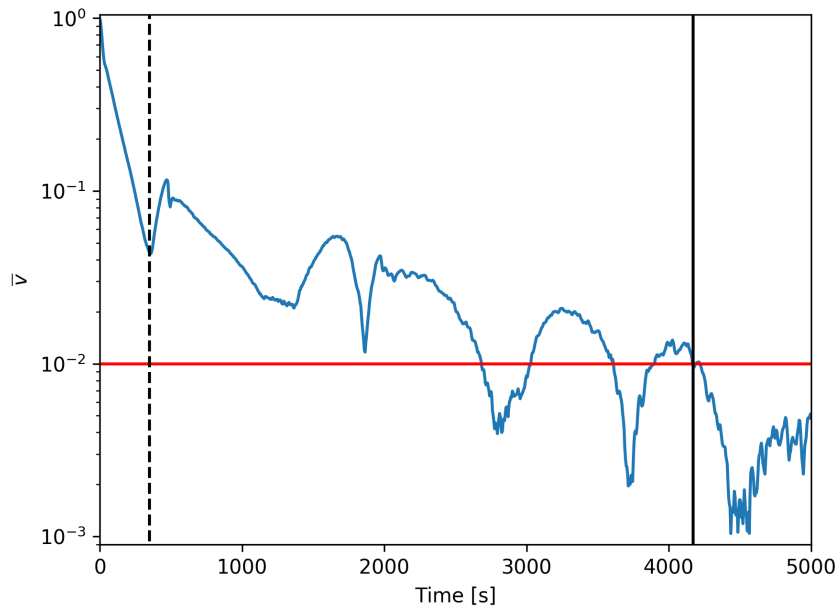


Figure 3: Temporal evolution of the average velocity of the impactor material over the initial impact velocity for the reference model ( $v_{imp} = v_e = 2.33 \text{ km s}^{-1}$ ,  $\delta_{crust} = 40 \text{ km}$  and  $R_{imp} = 50 \text{ km}$ ). The red line represents our arbitrary stop criterion (fixed to a velocity ratio of 1%). The black dashed line represents the time of formation of the transient crater. The black solid line represents the time at which we stopped our simulation.

510 and for monitoring its pressure/temperature evolution as well as the maximum  
 511 pressure/temperature over time. In our models, we consider a computational  
 512 domain that is large enough to limit the loss of material ejected during the  
 513 impact but not so extensive that it wastes computational time. Hence, the  
 514 conservation equations are solved in a 10,000 km  $\times$  2000 km domain. In the  
 515 following section, we define a reference case which also provides the basis for  
 516 subsequent discussion on the choice of the spatial resolution.

517

518 To choose the spatial resolution of our models, we ran the reference model  
 519 ( $R_{imp} = 50$  km,  $v_{imp} = v_e$  and  $\delta_{crust} = 40$  km) at different resolutions ranging  
 520 from 5 cells per projectile radius (cpr) to 50 cpr (which corresponds to a  
 521 grid size ranging from 10 km to 1 km respectively). iSALE enables to use a  
 522 non-uniform mesh with a resolution that is finer close to the impact site and  
 523 coarser in the antipodal region of the impacted Moon where deformation is  
 524 less significant. We investigated both uniform and non-uniform resolutions.  
 525 Non-uniform grids are associated to the index "ext" (i.e. cpr<sub>ext</sub>). For this  
 526 resolution test, we modeled the impact process up to 1000 s after the impact  
 527 and compared the final mass from our models to the initial mass of the impactor  
 528 ( $m_{imp,0} = \frac{4}{3}\pi R_{imp}^3 \rho_{imp}$  with  $\rho_{imp}$  the impactor density). The relative average  
 529 error is calculated as:

$$\epsilon(t) = \frac{(m_{imp}(t) - m_{imp,0})}{m_{imp,0}} \times 100 \quad (1)$$

530 with  $m_{imp}(t)$  the impactor mass from our models.  $\epsilon(t)$  is used as a criterion  
 531 for the selection of the resolution.

532

533 For very energetic impact cases, the impactor material can potentially be  
 534 buried deep within the target and may mix with the pre-existing lunar mantle.



535 As both the impactor and the lunar mantle are made of the same dunitic mate-  
 536 rial in our model, it may be difficult to differentiate between these two materials  
 537 using the Eulerian method. To overcome this problem, about one million La-  
 538 grangian tracer particles were implanted in the impactor to monitor its average  
 539 depth, pressure and temperature during the impact.

540

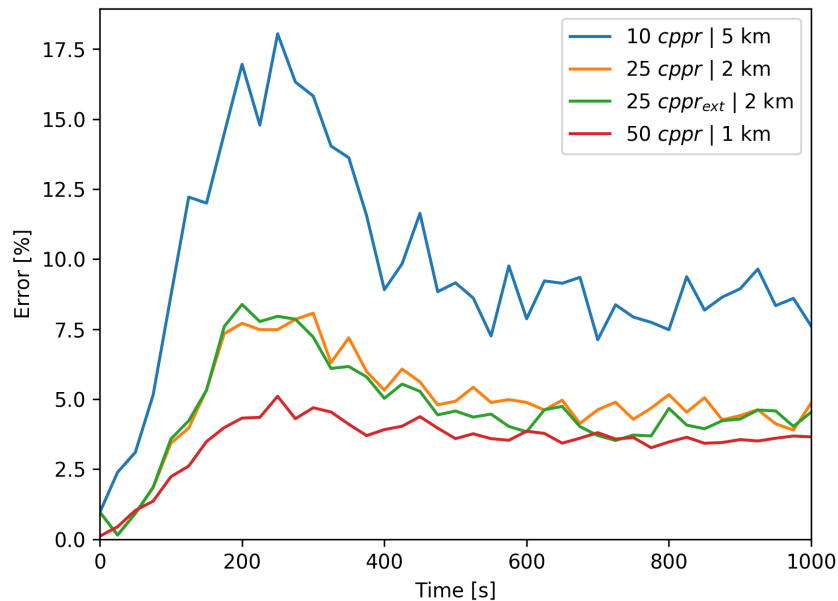


Figure 4: Time evolution of the mass error for the impactor ( $\epsilon(t)$ ) for different spatial resolutions (see Eq. 1). Cases with 10 cpr, 25 cpr and 50 cpr have a uniform mesh. For the case with 25 cpr<sub>ext</sub> the resolution is non-uniform with a resolution of 25 cpr close to the impact site and half this resolution in the other parts of the planet. We consider here the reference case with  $R_{imp} = 50$  km,  $v_{imp} = v_e = 2.33$  km s<sup>-1</sup> and  $\delta_{crust} = 40$  km.

541 Figure 4 shows the evolution of error for different resolutions as a function  
 542 of time and illustrates that the error is sensitive to the different impact stages.  
 543 During the contact/compression stage, the error is lower than 2.5% for all the  
 544 resolutions considered. During the excavation stage, which end up at 315 s,  
 545 the error increases up to a maximal value of  $\approx 8\%$  for the 25 cpr<sub>ext</sub> resolution.

546 Then the error decreases until the end of the excavation stage. During the mod-  
547 ification stage, the error varies slightly around a value of  $\approx 5\%$  for resolution of  
548 25 cppr. A relatively low error is obtained during the compression stage be-  
549 cause the impactor material experiences a density increase while concentrated  
550 in a few cells. On the contrary, during the vertical jet formation the material is  
551 dispersed in the surrounding cells and the error increases due to the numerical  
552 diffusion.

553

554 The computational time is also an important criterion to consider when  
555 choosing the resolution for our models. The computational time depends on the  
556 spatial resolution, the computational capacities, the size of the domain defined  
557 and the duration of the impact process. For the simulations detailed in our  
558 study, we used a computer with a SSD disk of 1 TB, memory 64 GB RAM and  
559 a processor Intel® Xeon(R) W-10855M CPU @ 2.80GHz  $\times 12$ . Figure 5 illus-  
560 trates the selection criteria with the computational time (blue) and the average  
561 error (red) as a function of the resolution in cppr. The average error decreases  
562 with an increase in resolution, while the computation time strongly increases  
563 with the increasing resolution. From 25 cppr we observe a significant increase  
564 in the calculation time along with a low error decrease. An increase in spatial  
565 resolution from 25 cppr to 50 cppr leads to only a 1.28% reduction in error, but  
566 a sevenfold increase in computation time. In Figure 5, we also show the results  
567 for cases with a non-uniform resolution which leads to a decrease in computation  
568 time of  $\approx 43\%$  while the mean error remains constant. To maintain a low error  
569 and a reasonable computational time we chose to run models using a resolution  
570 of 25 cppr with a non-uniform grid (see Tab. 3).

571

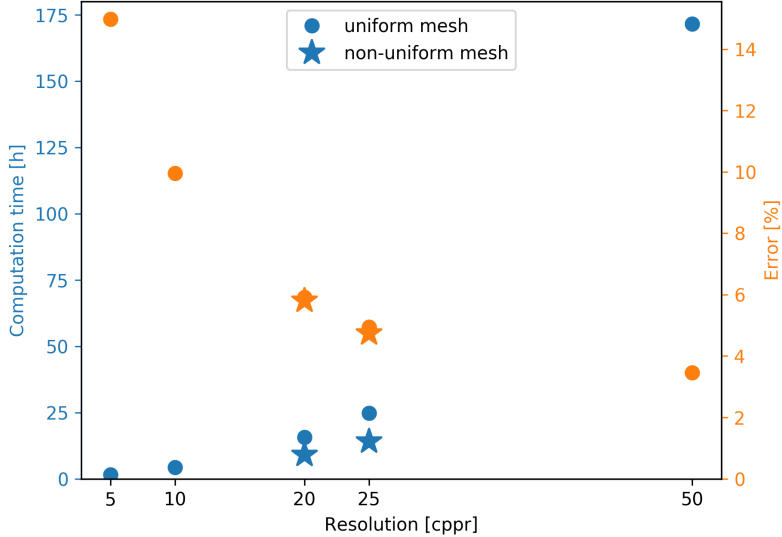


Figure 5: Computation time (blue) and error (orange) as a function of the resolution (See Eq. 1). Both values are obtained for the reference case  $t = 1000$  s after the impact. The circles represent the values obtained with a uniform mesh. The stars represent the values obtained with a non-uniform mesh. For the non-uniform meshes, the resolution (in cpr) is the resolution close to the impact site while the resolution far from the impact site is half the resolution close to the impact site. We consider here the reference case with  $R_{imp} = 50$  km,  $v_{imp} = v_e = 2.33$  km s $^{-1}$  km and  $\delta_{crust} = 40$ .

#### 572 4. Fate of the impactor: parametric study

573 Recently, *Jackson et al. (2023)* used numerical modeling to assess the im-  
574 pactor’s ability to penetrate through a lunar crust above a magma ocean. They  
575 explored a crustal range from 1 to 40 km, with impactor radii varying from 50  
576 m to 15 km and impact velocities from 3 to 15 km s $^{-1}$ . They derived scaling  
577 laws that relate the kinetic energy and the potential for lunar crust deforma-  
578 tion and penetration. In particular, they derive a scaling law that constrain the  
579 minimum kinetic energy required for the impactor to completely perforate the  
580 crust, leaving a hole in the crust that exposes the magma ocean at the surface.  
581 In the following section, we characterise the influence of the lunar crustal thick-  
582 ness ( $10 \leq \delta_{crust} \leq 80$  km), the impactor radius ( $25 \leq R_{imp} \leq 200$  km), and

583 the impact velocity ( $v_e \leq v_{imp} \leq 4v_e$ ) on the ability of an impact to bury ma-  
584 terial deep within a partially molten Moon. For that, we perform a parametric  
585 study of each parameter while keeping the other parameters fixed to appropriate  
586 values. In particular, we monitor the average depth of the impactor material  
587 using more than one million tracers for these three parameters and compare our  
588 results with the scaling law derived from *Jackson et al. (2023)*. According to  
589 *Jackson et al. (2023)*, we explore a range of parameters where the penetration of  
590 the impactor should be fully achieved. Videos relative to the following section  
591 illustrating the temperature and compositional evolutions during the impact are  
592 hosted within our repository file (see section 9 for access details).

#### 593 4.1. Influence of the lunar crustal thickness

594 We evaluate the influence of the target crustal thickness ( $\delta_{crust}$ ) on the im-  
595 pactor penetration depth. According to *Wieczorek et al. (2013)*, the lunar crust  
596 currently has an average thickness of  $\approx 40$  km with a maximum thickness of 60  
597 km. To identify the influence of the crustal thickness, we consider a range of  
598 crustal thickness values with an upper bound value larger than the lunar crustal  
599 thickness. Here, we fix the impactor radius to 50 km and the impact velocity  
600 to  $v_{imp} = v_e$ .

601  
602 Figure 6 (top panel) shows the depth reached by the impactor material as  
603 a function of the crustal thickness. It illustrates the depth at the end of the  
604 simulation (see Subsection 3). It shows the configuration of the impactor mate-  
605 rial when the cratering flow ceases and buoyancy forces dominate, leading to a  
606 further stage of mechanical adjustment by turbulent entrainment (not modeled  
607 here). In this figure, we consider two depths: (1) the average depth which is  
608 calculated as the average depth of all the tracers and (2) the maximum depth  
609 which represents the depth of the deepest tracer. The difference between the

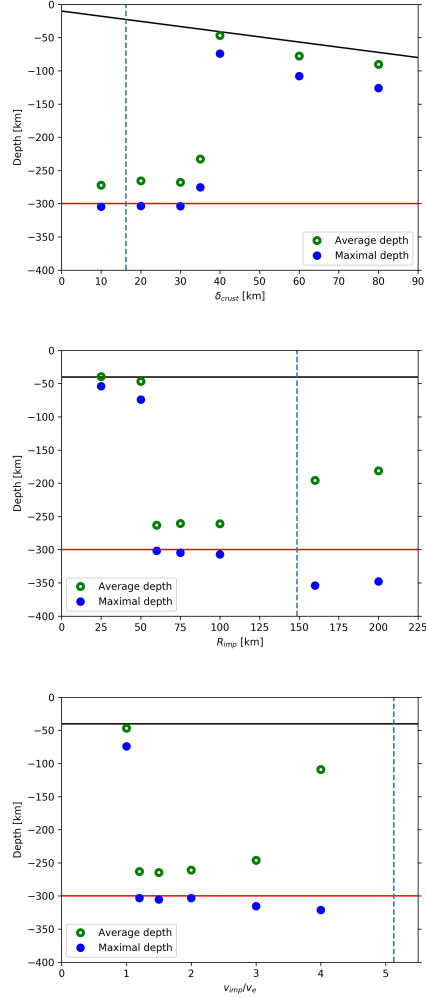


Figure 6: Final average (green symbols) and maximal (blue symbols) depth of the impactor material (top panel) as a function of the planet crustal thickness ( $\delta_{crust}$ ) for an impactor of radius 50 km with an impact velocity of  $v_{imp} = v_e = 2.33 \text{ km s}^{-1}$ ; (middle panel) as a function of the impactor radius  $R_{imp}$  for a crustal thickness of 40 km with an impact velocity of  $v_{imp} = v_e = 2.33 \text{ km s}^{-1}$ ; (bottom panel) as a function of the impact velocity ( $v_{imp}$ ) over the escape velocity ( $v_e$ ) for a crustal thickness of 40 km and an impact radius of 50 km. The red solid line corresponds to the boundary between the magma ocean and the solid mantle. The black solid line illustrates the crustal thickness of each model. The dashed blue line is the partial-complete penetration limit from *Jackson et al. (2023)*

610 average depth and the maximal depth illustrates the deformation and fragmen-  
611 tation of the impactor at the end of the impact process. We note here that we  
612 do not consider any fragmentation model in our calculations (*Wiggins et al.*,  
613 2019). Figure 6 (top panel) also shows the thickness of the lunar crust. If the  
614 average depth is above this line, the impactor does not penetrate the crust,  
615 while below it, the impactor ends up within the magma ocean. Finally, the  
616 boundary between the magma ocean and the lower mantle is also plotted to  
617 illustrate the capacity of the impactor material to deform the deep mantle and  
618 bury the impactor material deeper than the magma ocean.

619

620 Figure 6 (top panel) shows that increasing the thickness of the crust de-  
621 creases the ability of the impactor material to reach great depths within the  
622 lunar interior. For  $\delta_{crust} < 30$  km, the impactor easily penetrates the crust  
623 and reaches the bottom of the magma ocean. Even if the solid mantle beneath  
624 the magma ocean acts as an efficient barrier limiting the penetration of the im-  
625 pactor, for  $\delta_{crust} < 30$  km, we observe some tracers at depths larger than 300  
626 km, meaning that the impactor has slightly deformed the boundary between the  
627 solid mantle and the magma ocean. For  $\delta_{crust} \geq 40$  km, the impactor can still  
628 penetrate within the magma ocean but the crust strongly limits the penetration  
629 of the impactor. For these cases, in average, the impactor material finally lies  
630 beneath the crust and the final depth of the buried material increases as the  
631 crustal thickness increases. Its final position seems to be influenced by the size  
632 of the jet (see section 3) which decreases with an increase of the crust thickness.

633

#### 634 4.2. Influence of the impactor radius

635 The Moon likely experienced several very large impacts during its early evo-  
636 lution. These large impacts probably involved impactors with radii up to 300 km

637 (*Bottke et al.*, 2010; *Morbidelli et al.*, 2018) while the Moon was still partially  
638 molten. Here, we evaluate the ability of a large impactor ( $25 \leq R_{imp} \leq 200$  km)  
639 to penetrate through the solid crust and down into the underlying magma ocean.  
640 Here, we fix the crustal thickness to 40 km and the impact velocity to  $v_{imp} = v_e$ .

641

642 Figure 6 (middle panel) shows the depth reached by the impactor material  
643 as a function of the impactor radius. For  $R_{imp} \leq 50$  km, the impactor pene-  
644 trates the crust but the post-excavation processes and the jet dynamics tends  
645 to maintain the impactor material at shallow depths within the magma ocean  
646 (Fig. 2). For  $50 < R_{imp} < 160$  km, the impactor is large enough to penetrate  
647 the crust and reaches the bottom of the magma ocean. For impactors radii  
648 larger than 160 km, a larger kinetic energy is involved which leads to greater  
649 deformation and fragmentation. For these cases, the impactor reaches the top  
650 of the solid lower lunar mantle, which acts as a mechanical barrier. Then, the  
651 impactor material spreads out within the magma ocean over a radial extent of  
652 more than 600 km and a thickness of 40 km. Ultimately, a large vertical jet is  
653 generated by the impact induced rebound which reduces the final average depth  
654 of the impactor material.

655

#### 656 4.3. Influence of the impactor velocity

657 During the early stages of an impact, the kinetic energy of the impactor is  
658 rapidly transferred to the target body (*Melosh*, 1989). The available kinetic  
659 energy depends mainly on two parameters: the impactor mass (related to its  
660 radius, see section above) and the square of the impactor velocity. Hence, the  
661 impactor velocity should be a major parameter in determining the ability of  
662 the impactor to penetrate deep into the lunar interior. On the Moon, impacts  
663 can occur at a wide range of velocities. The minimum impact velocity is in the

664 order of the escape velocity of the Moon ( $v_e$ ). Between 4.5 and 4.15 Ga, the  
 665 median impactor velocity on the Moon is estimated to  $9 \text{ km s}^{-1}$  (*Bottke et al.*,  
 666 2012) and to have increased later on (*Marchi et al.*, 2012). The hypervelocity  
 667 impacts ( $\geq 10 \text{ km s}^{-1}$ ) induced by giant planet migrations are likely to have  
 668 occurred during the late accretion (after 4.15 Ga) (*Marchi et al.*, 2012). The  
 669 impact velocity can have a significant influence on the efficiency of the impact  
 670 accretion (*Zhu et al.*, 2019a), the fragmentation of the impactor (*Kendall and*  
 671 *Melosh*, 2016) and the melt fraction of the impactor material after the impact  
 672 (*Pierazzo et al.*, 1997). In this section, we consider impact velocities ranging  
 673 between 1 and 4 times the escape velocity of the impacted Moon. We fix here  
 674 the crustal thickness to 40 km and the impactor radius to 50 km.

675

676 Figure 6 (bottom panel) shows that for all impact velocities considered here,  
 677 the impactor penetrates the 40 km-thick crust. For  $v_{imp} = v_e$  (corresponding to  
 678 our reference model), the impactor material settles at shallow depth below the  
 679 crust. However, for  $v_{imp}/v_e > 1$ , the impactor reaches the bottom of the magma  
 680 ocean. In these cases, the penetration efficiency is limited by the solid lower  
 681 mantle strength. Impacts with velocities larger than  $3 v_e$  deform the upper part  
 682 of the solid mantle underlying the magma ocean below the impact site, enabling  
 683 impactor material to penetrate below 300 km depths. For the highest impact  
 684 velocities considered here, a large vertical jet is induced during the modification  
 685 stage. The height of this jet increases with the impact velocity. This height  
 686 can exceed 700 km above the lunar crust for the highest impact velocities. The  
 687 collapse of this massive jet can transport a significant impactor fragments to-  
 688 wards the planet surface or redistribute the fragments within the magma ocean  
 689 over a wide range of depths. As a consequence, unlike the maximal depth of the  
 690 impactor material, the average depth of the impactor material decreases as the



691 velocity increases.

692

#### 693 4.4. Summary

694 For all parameter values used in our study, the impactors penetrate through  
695 the crust. For cases with a thick crust, a small impactor radius and/or a low  
696 impact velocity, the impactor material is partially buried below the lunar crust.  
697 For more energetic cases, the impactor material penetrates into the magma  
698 ocean and can be re-entrained by movements within the convecting magma  
699 ocean (*Maas et al.*, 2021). Such a secondary process is not considered in our  
700 models. *Jackson et al.* (2023) derived a scaling law characterising the impact  
701 energy threshold above which the impactor material penetrates totally within a  
702 lunar magma ocean as a function of the thickness crust. Hence, assuming a fixed  
703 impact velocity (or a fixed impactor radius), one can obtain the corresponding  
704 impactor radius (or the impact velocity) threshold separating complete and  
705 partial penetration regimes. We illustrate these values in Figure 6 (with blue  
706 dashed lines). Figure 6 (top panel) shows that the partial-complete penetration  
707 threshold obtained from our models ranges between 30 and 40 km for the crustal  
708 thickness. This value is larger but still in agreement with the threshold thickness  
709 of 19 km derived from *Jackson et al.* (2023) scaling law. In figure 6 (middle  
710 panel), the threshold for impactor radius obtained from our models is decreased  
711 to a radius of  $\sim 60$  km compared to the value of 149 km from *Jackson et al.*  
712 (2023). Finally, figure 6 (bottom panel) shows an impactor velocity threshold  
713 of  $1.1 v_e$  which is again lower than the  $5.1 v_e$  derived from *Jackson et al.* (2023)  
714 scaling law. These results illustrate that, in our models, less impact kinetic  
715 energy is required than expected to penetrate completely through the lunar  
716 crust. This difference can be attributed to differences in the range of impactor  
717 radius (larger in our study) or to the strength properties of both the crustal and

718 impactor materials.

## 719 **5. P, T and melt evolution of the impactor**

720 During the impact process, the impactor material is subjected to major de-  
721 formation as well as temperature and pressure modifications. Depending on  
722 the impact parameters and on the time evolution of the impact process, the im-  
723 pactor material can locally or globally reach the P-T conditions at which partial  
724 to complete melting occurs. To investigate whether melting is likely to occur at  
725 the surface or deep in the magma ocean, we monitor the P-T evolution of the  
726 impactor material at two specific times: the end of the excavation stage and the  
727 end of the modification stage (i.e. the end of our simulations).

728

### 729 *5.1. Reference case*

730 We monitored the P-T evolutions of the impactor material using one million  
731 Lagrangian tracers. Figure 7 shows the distribution of the maximum pressure  
732 and temperature reached by the population of tracers. These results are derived  
733 from the reference case described in Section 2 with  $R_{imp} = 50$  km,  $v_{imp} = v_e$   
734 and  $\delta_{crust} = 40$  km (see Figure 2). Two sets of histograms are illustrated in  
735 Figure 7 corresponding to the two key times specified above.

736

737 The contact/compression stage has the greatest pressure increase recorded  
738 in this reference case (up to 29 GPa). At the end of the excavation stage, fol-  
739 lowing the contact/compression stage, the transient crater collapses followed by  
740 the formation of a vertical magma jet. During this second stage, which ter-  
741 minates at the end of the modification stage, the impactor experiences a new  
742 pressure increase, but lower than the pressure increase occurring during the  
743 contact/compression stage. This supplementary pressure increase is induced by

744 the inflow of material within the crater. Figure 7 (left) illustrates the maxi-  
745 mum pressure increase experienced by the tracers during this impact process.  
746 The distribution of pressure at the end of the excavation stage is right-skewed  
747 (shifted towards a low pressure increase), while the distribution at the end of  
748 the modification stage is slightly shifted to the right. The pressure increase at  
749 the end of the excavation stage lies between 0 and 29 GPa with an average value  
750 of 5.8 GPa. At the end of the modification stage, the pressure increase ranges  
751 from 0.3 to 29 GPa with an average value of 6.1 GPa.

752

753 During the impact process, a significant fraction of the impact energy is  
754 dissipated as heat within both the impacted body and the impactor. The tem-  
755 perature increase within the impactor material is mainly located in the region  
756 in contact with the proto-planet (i.e. the lower surface of the impactor). The  
757 temperature increase is moderate in our reference case as the impact velocity  
758 is only  $2.33 \text{ km s}^{-1}$ . Figure 7 (right) illustrates the maximum temperature  
759 increase experienced by the tracers during the impact process. It shows that  
760 the populations before and after the modification stage are very similar. The  
761 distributions for the temperature increase at the end of the excavation stage  
762 and at the end of the modification stage are both right-skewed. During the  
763 modification stage, the crater walls collapse and the temperature may increase  
764 in the upper part of the impactor. The temperature increase at the end of the  
765 excavation stage lies between 188 and 1656 K with an average value of 605 K,  
766 whereas at the end of the modification stage the temperature increase is between  
767 371 and 2196 K with an average value of 727 K.

768

769 From this data, we calculated the melt fraction  $\phi_{melt}$  of the impactor mate-  
770 rial before and after the modification stage using the melting curves (solidus and

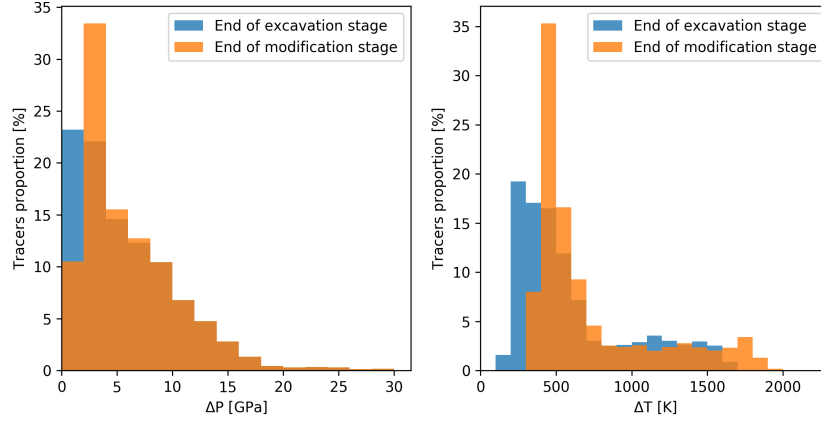


Figure 7: Maximum pressure increase (left) and maximum temperature increase (right) for each tracer in the reference model with  $R_{imp} = 50$  km,  $v_{imp} = 1 v_e$  and  $\delta_{crust} = 40$  km. Blue histograms corresponds to the end of the excavation stage ( $t=310$  s), while orange histograms are obtained at the end of the modification stage (i.e. the end of the simulation with  $t=4170$  s).

771 liquidus) derived for chondritic material from *Andrault et al. (2018)* for  $P \leq 24$   
 772 GPa:

773

$$T_{sol} = 1373 \left( \frac{P}{0.82 \times 10^9} + 1 \right)^{(1/6.94)} \quad (2a)$$

$$T_{liq} = 1983.4 \left( \frac{P}{6.48 \times 10^9} + 1 \right)^{(1/5.35)} \quad (2b)$$

774 where  $T_{sol}$  is the solidus temperature and  $T_{liq}$  is the liquidus temperature.

775 For  $P \geq 24$  GPa, we used melting curves derived from *Andrault et al. (2011)*:

$$T_{sol} = 1334.5 \left( \frac{P}{9.63 \times 10^9} + 1 \right)^{(1/2.41)} \quad (2c)$$

$$T_{liq} = 1862 \left( \frac{P}{21.15 \times 10^9} + 1 \right)^{(1/2.15)} \quad (2d)$$

776

777

778       Calculating the melt fraction for each tracer at each time step of our sim-  
779       ulations, we obtained the total melt fraction within the impactor material at  
780       the end of the excavation and modification stages. For the reference case, the  
781       melt fraction is 3.87% at the end of the excavation stage and 10.17% at the end  
782       of the modification stage. Thus we can infer that, under the conditions of this  
783       reference case, the impactor melting is not significant. We then carried out a  
784       parametric study to characterise the influence of the following three parameters  
785       ( $\delta_{crust}$ ,  $R_{imp}$  and  $v_{imp}$ ) on the ability of the impactor material to melt during  
786       the impact process. The pressure and temperature increase distributions from  
787       our models are similar to the distributions illustrated in Figure 7 (left). His-  
788       tograms relative to the following section are hosted within our repository file  
789       (see section 9 for access details)

790

## 791 *5.2. Parametric study*

792       In Figure 8 (top panel), we illustrate the average melt fraction of the im-  
793       pactor as a function of the crustal thickness ( $\delta_{crust}$ ) at the end of both excavation  
794       and modification stages. We consider here crustal thicknesses ranging from 10  
795       to 80 km while we fixed the impactor radius to 50 km and the impact velocity  
796       to  $v_e$ . Figure 8 (top panel) shows that the melt fraction is nearly twice larger at  
797       the end of the modification stage than at the end of the excavation stage. This  
798       means that the two stages have the same contribution to the melting fraction of  
799       the impactor. However, the thickness of the crust has a small influence on the  
800       fraction of impactor material that experiences melt. The melt fraction at the  
801       end of each simulation is lower than 16% for our whole range of  $\delta_{crust}$  values.  
802       Figure 8 (top panel) illustrates that the total melt fraction decreases with in-

803 creasing values of  $\delta_{crust}$  from 15.6 % (for  $\delta_{crust} = 20$  km) to 1.2% (for  $\delta_{crust} = 80$   
 804 km). For the current estimation of the lunar thickness (i.e.  $\delta_{crust} \approx 40$  km),  
 805 the melt fraction of the impactor at the end of the modification stage is only  
 806  $\approx 10$  % and the impactor material is buried in the magma ocean in a mostly  
 807 solid state.

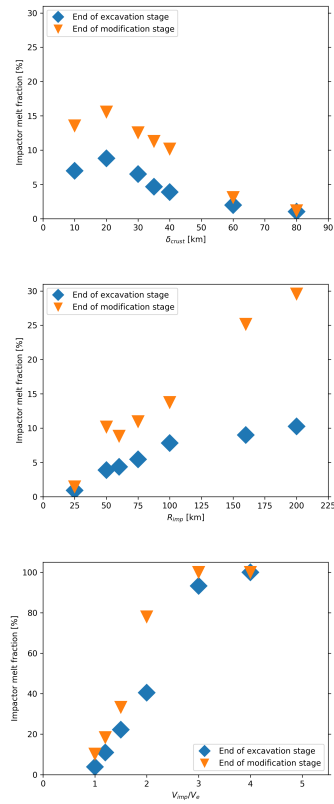


Figure 8: Average melt fraction of the impactor as a function of the crustal thickness (top panel), the impactor radius (middle panel) and the impact velocity (bottom panel). Results are obtained at the end of the excavation stage (blue) and at the end of the modification stage (orange). The material above the chondritic solidus is considered as partially molten up to the liquidus, above which it becomes fully molten.

808 We also investigated the melt fraction evolution of the impactor material as  
 809 a function of the impactor radius (Figure 8, middle panel). We tested different  
 810 values for the impactor radius ranging from 25 km to 200 km and we fixed the

811 crustal thickness to 40 km and the impact velocity to  $v_e$ . Figure 8 (middle panel)  
812 shows the average melt fraction of the impactor as a function of the impactor  
813 radius at the end of both the excavation and modification stages. In both cases,  
814 the impactor melt fraction increases with the impactor radius. At the end of  
815 the excavation stage, the average impactor melt fraction increases from 0.9 %  
816 to 10.2% and at the end of the modification stage, the increase ranges between  
817 1.43 % and 29.6%. Figure 8 (middle panel) also shows that the increase is  
818 more pronounced at the end of the modification stage. This means that most of  
819 the melting process of the impactor occurs late in the impact process when the  
820 impactor material is likely to be buried below the crust within the magma ocean.

821

822 Finally, we monitored the melt fraction evolution of the impactor material  
823 as a function of the impactor velocity considering  $v_{imp}$  values between  $v_e$  and 4  
824  $v_e$  (Figure 8, bottom panel). Here, we fixed the crustal thickness to 40 km and  
825 the impactor radius to 50 km. Figure 8 (bottom panel) shows the average melt  
826 fraction of the impactor as a function of the impact velocity at the end of the  
827 excavation and the modification stage. Our results illustrate that the impactor  
828 velocity strongly governs the melting of the impactor material. For the lowest  
829 impact velocity, 3.9 % of the impactor has experienced melting at the end of  
830 the excavation stage and 10.2 % at the end of the modification stage. These  
831 values increase to 100% at the end of both stages for  $v_{imp} = 4v_e$ . Between  $v_e$   
832 to  $3 v_e$ , the melt fraction at the end of the excavation stage increases linearly  
833 from 3.9 % to 93.3 %. These results illustrate the sensitivity of our melting  
834 models to the impact velocity. Above a critical value  $v_{imp} = 4v_e$ , the impactor  
835 material is fully molten before reaching the magma ocean. The only way to  
836 envision impactor material penetrating the magma ocean without experiencing  
837 significant melt is to consider low impactor velocities close to or slightly larger

838 than the lunar escape velocity.

## 839 **6. Impactor degassing**

840 In the previous section, we have estimated the pressure and temperature con-  
841 ditions within the impactor during an impact process involving a molten layer  
842 below the Moon’s crust. In the following section we make an attempt to quan-  
843 tify the dehydration from the impactor’s material and the water partitioning  
844 between the Moon’s surface and its interior.

### 845 *6.1. Impact degassing criterion*

846 The dehydration enhanced by meteoritic impacts is a complex process. Ac-  
847 cording to experimental studies (*Lange and Ahrens, 1982; Nakato et al., 2008;*  
848 *Nozaki et al., 2006*), the dehydration of chondritic material is related to the  
849 destabilisation of phyllosilicates and should initiate at a temperature of  $\sim 873$   
850 K. *Nakato et al. (2008)* showed that phyllosilicates need to be heated at 873 K  
851 for a few days to be partly dehydrated. At 1173 K the dehydration of phyl-  
852 losilicates is more efficient and occurs in less than one hour. The dehydra-  
853 tion process is also pressure dependent. If the dehydration reaction is weakly  
854 pressure-dependent below a peak shock pressure of 21 GPa, *Sekine et al. (2015)*  
855 showed that it becomes significant at pressures ranging between 21 and 60 GPa.  
856 Their experiments suggest a full dehydration of antigorite at  $T = 1073$  K. At  
857 relatively higher temperatures (up to 1500 K) forsterite dust grains are able to  
858 retain chemisorbed water at their surfaces (*King et al., 2010*). Like the CI chon-  
859 drites, the CM chondrites contain mainly serpentine minerals. Hence the CM  
860 chondrites can be considered as an analogue to the dehydration of serpentines in  
861 chondrites. *Thompson et al. (2021)* experimentally heated CM chondrites from  
862 475 to 1475 K during a period of 5 hours. They showed that at  $T = 973$  K, the  
863 partial pressure of the vaporised water decreases, but the water is still part of



864 the majority of the volatile compounds degassed at 973 K for more than two  
865 hours. Hence, it appears that the dehydration process enhanced by an impact  
866 strongly depends on the temperature and pressure conditions related to the im-  
867 pact process itself. But it also strongly depends on the chemical composition of  
868 the material involved, the main carrier phases of water, and the duration of the  
869 destabilisation process.

870

871 To understand the origin of hydrous minerals on the surface of asteroids,  
872 *Wakita and Genda* (2019) developed numerical impact models assuming a de-  
873 hydration temperature of 873 K. This temperature corresponds to an absolute  
874 specific entropy of dehydration  $S_{dehydr} = 3.2143 \text{ kJ K}^{-1} \text{ kg}^{-1}$ . As illustrated  
875 in the previous sections, the impact processes occur in a very short in time.  
876 In our models, between the high pressures generated by the contact and com-  
877 pression stages and the modification stage where the crater walls collapse, the  
878 impactor material may be destabilised for a period of maximum 300 s. Hence  
879 it is likely that a portion of the impactor material heated at 873 K during a  
880 few minutes would not fully dehydrate. Moreover, we have illustrated in the  
881 previous sections that the peak pressure increase was strongly dependent of the  
882 impact velocity. For  $v_{imp} = v_e$  and  $R_{imp} = 50 \text{ km}$  the pressure increase reaches  
883 a maximum value of 29 GPa while for  $v_{imp} = 4v_e$ , the maximum pressure in-  
884 crease may reach a value of 121.4 GPa. As a consequence and in agreement  
885 with *Sekine et al.* (2015), it seems relevant to consider a dehydration criterion  
886 that is both a function of pressure and temperature.

887

888 Figure 9 compares the fraction of the impactor material that is dehydrated  
889 for the two discussed criterion: (1) a dehydration related to the melt frac-  
890 tion of chondritic material (from solidus and liquidus) and (2) the dehydration

891 temperature of 873 K proposed by *Wakita and Genda* (2019). For these two  
892 dehydration models, we compare the fraction of the impactor dehydrated at the  
893 end of the excavation stage (that is likely to degas within the lunar atmosphere)  
894 and at the end of the modification stage (that is likely to degas within the lunar  
895 mantle). The material that is not degassed at the end of the excavation stage  
896 should also contribute to the water content of the lunar mantle. Figure 9 shows  
897 that the fraction of the impactor material that is likely to experience dehydra-  
898 tion is larger when considering a dehydration temperature of 873 K instead of  
899 the molten volume fraction. We observe that the difference between the two  
900 dehydration models can represent nearly 70% in particular for the cases with  
901  $v_e < v_{imp} < 2v_e$  (Figure 9, bottom) or for the cases with  $R_{imp} > 150$  km (Figure  
902 9, middle). The conditions for full dehydration are more difficult to reach as-  
903 suming the melt fraction criterion. Indeed, in this case and at ambient pressure,  
904 dehydration should initiate at  $T=1373$  K and full dehydration should occur at  
905  $T=1983$  K. Hence the melt fraction criterion represents a lower bound of the  
906 dehydration induced by impact while the 873 K dehydration temperature crite-  
907 rion represents an upper bound. Hence using the melt fraction of the impactor  
908 material as a proxy of the impactor dehydration likely overestimates the amount  
909 of water that can be buried within the magma ocean while the model proposed  
910 by *Wakita and Genda* (2019) likely overestimates the amount of surface water  
911 degassed by impact. Finally, we note that once the impactor is buried within  
912 the magma ocean, the pressure overlying the impacted material is high enough  
913 for its water to be dissolved into the magma. Hence, all the material that is not  
914 dehydrated during the impact will potentially contribute to the water content  
915 of the Moon’s mantle.

916

917 Figure 9 also shows that both excavation and modification stages contribute

918 to the dehydration with similar contributions. For example, for  $R_{imp} = 160$   
 919 km (Figure 9, middle) and assuming the melt fraction criterion, 9% of the  
 920 impactor material is dehydrated during the excavation stage and 16% is dehy-  
 921 drated during the modification stage. This feature is also visible assuming a 873  
 922 K dehydration temperature. For  $R_{imp} = 160$  km, 55% of the impactor material  
 923 is dehydrated during the excavation stage and 40% is dehydrated during the  
 924 modification stage. However, Figure 9 (top and middle) shows that the rela-  
 925 tive contribution from the excavation stage increases with the crust thickness  
 926 while it decreases with the radius of the impactor. Hence, assuming that the  
 927 excavation stage contributes to the degassing within the lunar atmosphere and  
 928 the modification stage contributes to the hydration of the lunar interior, large  
 929 impacts on a thin lunar crust should favour degassing within the lunar magma  
 930 ocean.

931

932 Finally, Figure 9 (bottom) shows that the impact velocity plays a key role in  
 933 the degassing process. For  $v_{imp} \geq 3v_e$ , the impactor material is fully dehydrated  
 934 during the excavation stage independently of the dehydration model. For these  
 935 large impact velocities, all the water content from the impactor should degas  
 936 at the surface of the Moon and contribute to its early atmosphere composition.  
 937 However, for  $1 < v_{imp} < 3v_e$ , Figure 9 (bottom) shows that it is possible for  
 938 the impactor material to penetrate the crust without experiencing a full dehy-  
 939 dration. For example, for  $v_{imp} = v_e$  and considering the melt fraction criterion,  
 940 only 4% of the impactor is dehydrated at the end of the modification stage  
 941 and 10% at the end of the modification stage. When considering a dehydration  
 942 temperature of 873 K, these two values increase to 35% and 42% respectively.  
 943 This means that, depending on the dehydration model, between 65% and 96%  
 944 of the water content of the impactor is likely to contribute to the water budget

945 of the lunar interior for low velocity impacts (with  $v_{imp} = v_e$ ).

946

947

## 948 *6.2. Contribution of large impacts to the lunar water content*

949 The mass of material accreted within the Moon’s mantle after the Moon  
950 forming impact has been constrained using highly siderophile elements and the  
951 Re-Os systematics. *Day et al.* (2007) calculated a contribution of 0.02% of CI-  
952 type chondrite to the lunar mass which corresponds to  $1.5 \times 10^{19}$  kg. More  
953 recently it has been estimated that nearly  $2.4 \times 10^{20}$  kg of chondritic material  
954 have impacted the Moon during the lunar magma ocean stage (between 4.3 and  
955 4.5 Gyr) to explain the lunar HSE budget (*Zhu et al.*, 2019a). The contribution  
956 from impactors with radii larger than 25 km can be estimated to  $\approx 10\%$  of the  
957 total impactor mass (i.e.  $2.4 \times 10^{19}$  kg) (*Zhu et al.*, 2019a). From our models,  
958 we can estimate the fraction of a large impactor that is buried beneath the solid  
959 crust without experiencing melting during the impact process. This provides  
960 an estimation of the mass of water that can be buried within the LMO during  
961 the late lunar accretion.

962

963 We have seen in the previous sections that, in our range of parameters, a  
964 large impactor is likely to penetrate the crust and to be buried within the under-  
965 lying magma ocean. If we focus on the early stages of the lunar solidification, the  
966 crustal thickness has increased with time up to its current thickness  $\delta_{crust} = 40$   
967 km (*Wieczorek et al.*, 2013). However our results show a weak influence of the  
968 lunar crust thickness on the dehydration fraction for  $\delta_{crust}$  ranging between 10  
969 and 40 km. For this range of crustal thicknesses, the dehydration fraction re-  
970 mains below a value of 10% (Figure 9, top). This feature also stands for the  
971 impactor radius. Degassing increases with the size of the impactor but the de-

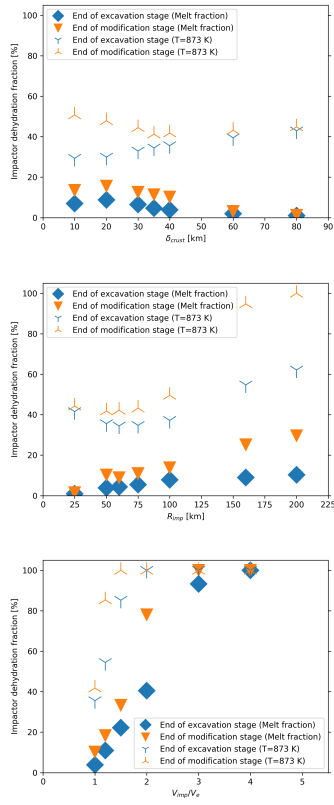


Figure 9: Fraction of the impactor experiencing dehydration as a function of the crustal thickness (top panel), the impactor radius (middle panel) and the impact velocity (bottom panel). The blue symbols represent the results obtained at the end of the excavation stage assuming the melt fraction criterion (squares) and the dehydration temperature of 873 K (crosses). The orange symbols represent the results obtained at the end of the modification stage using the melt fraction criterion (triangles) and the dehydration temperature of 873 K (crosses).

972 hydration fraction remains below 10% when the impactor reaches the magma  
973 ocean for  $25 < R_{imp} < 200$  km (Figure 9, middle). However, our models show  
974 that (1) for  $v_{imp} = 4v_e$ , the impactor material is significantly molten and de-  
975 gassed before reaching the lunar interior and (2) for  $v_{imp} = v_e$ , the impactor  
976 material is mostly solid and remains hydrated when until buried below the lunar  
977 crust (Figure 9, bottom).

978

979 The potential water contribution from large and low velocity impacts within  
980 the LMO depends on the water content of the impactor. The hydrogen isotopic  
981 composition of water in lunar samples is similar to carbonaceous chondrites  
982 and addition of CI-type material has been commonly suggested (*Tartèse et al.*,  
983 2013; *Füri et al.*, 2014). Chondrites from this group are the most enriched in  
984 terms of volatile elements (*Hauri et al.*, 2015) with a water content of 8.5 wt%  
985 for the average water concentration of a CI-type chondrite (*Piani et al.*, 2020).  
986 We consider this composition as our water-rich case. Enstatite chondrites (EC)  
987 represent the most depleted hydrogen end-member but the new calculation of  
988 H content by *Piani et al.* (2020) gives water contents ranging from 0.08 to 0.54  
989 wt % (i.e. 800 to 5400 ppm). Then EC are not devoid of water as previously  
990 suggested and could have contributed to the endogenous lunar volatile contents  
991 during the late accretion stage. We consider this composition as our water-poor  
992 case. If 10% of the total impacted mass (i.e.  $2.4 \times 10^{20}$  kg, (*Zhu et al.*, 2019a))  
993 accretes on the Moon from impactors with  $R_{imp} \geq 25$  km and that 90% of this  
994 mass is trapped within the magma ocean, it represents a potential source of  
995  $1.9 \times 10^{18}$  kg of water (for the water-rich case) and  $1.8 \times 10^{16}$  to  $1.22 \times 10^{17}$  kg  
996 of water (for the water-poor case) in this molten reservoir. The water contri-  
997 bution can be lower for differentiated impactors (divided by a factor 3 for EC  
998 aubrite)

1000 Assuming a 100 km-thick lunar magma ocean below a 40 km-crust, the mass  
 1001 of the magma ocean is  $1.14 \times 10^{22}$  kg. If the magma ocean is initially depleted in  
 1002 water, the water content after this late accretive event could vary between 168.7  
 1003 ppm (for the water-rich case) and from 1.6 ppm (for the water-poor case with  
 1004 water concentration of 0.08 wt%) to 10.7 ppm (for the water-poor case with wa-  
 1005 ter concentration of 0.54 wt%) for EC. For the water-poor case, we obtain water  
 1006 concentrations consistent with estimations proposed for the lunar mantle source  
 1007 (i.e. 1.2 to 74 ppm of water, (*McCubbin et al.*, 2023)). The water concentration  
 1008 calculated using the water-rich case exceeds the water concentration estimated  
 1009 in the bulk silicate Moon from lunar sample measurements. Our estimations  
 1010 can easily be reconciled with the measurements if only 5% of the total impacted  
 1011 mass comes from impactors with  $R_{imp} \geq 25km$ .

1012

1013 Finally, for the consistency of our scenario, two features have to be accounted  
 1014 for: (1) the ability of the accreted material to be fully homogenised within the  
 1015 LMO or (2) the ability of this material to be distributed within the mantle  
 1016 so that it can be sampled statistically. We note that, in this scenario, the  
 1017 total mass accreted from impactors with  $R_{imp} \geq 25km$  represents the mass  
 1018 of  $\approx 100$  impactors with  $R_{imp} = 25$  km or a single impactor with  $R_{imp} =$   
 1019 125 km. If full chemical homogenisation is likely in the LMO then the water  
 1020 content can be explained by a single 125 km radius impact made of water-  
 1021 rich material. However, such a chemical homogenisation within a thin LMO  
 1022 is difficult to imagine because of the lack of vigorous convection. Moreover,  
 1023 such a scenario would be difficult to reconcile with the apparent water content  
 1024 heterogeneities found in the lunar samples. Complementary models are needed  
 1025 to explore this point which is beyond the scope of our study. A late delivery

1026 of water by 100 large water-poor impacts into the lunar interior would help  
1027 reconcile the different water estimates because such a process would probably  
1028 favour large chemical heterogeneities in the lunar interior. Moreover, to favor  
1029 a heterogenous water content in the lunar mantle, it is likely that impactors of  
1030 different origins contributed to the water budget of the lunar mantle during its  
1031 late accretion.

## 1032 **7. Discussion**

1033 Our models represent a first attempt to constrain the mass of impactor  
1034 material that could be accreted deep within the early Moon following a large  
1035 impact and in particular how the impactor material could be trapped within the  
1036 residual LMO located below a thick crust. As a first step, several simplifying  
1037 assumptions have been made. Among these simplifications, not accounting for  
1038 the porosity effects is likely to be the most important assumption since large  
1039 range of values may be found among the different types of chondrites. For  
1040 example, measurements made on CI chondrites show porosity values of  $\approx 30\%$   
1041 (*Ostrowski and Bryson, 2019*). However, *Davison et al. (2010)* show that it is  
1042 easier to melt porous impactors. Not accounting for the porosity may lead to an  
1043 underestimation of the fraction of the molten material during an impact. This  
1044 effect is limited in the cases of our study since we consider relatively low impact  
1045 velocities. However, in future models, when considering larger and probably  
1046 more plausible impact velocities, the porosity effects have to be accounted for.

1047

1048 Once the impactor material is buried within the magma ocean, its capacity  
1049 to be homogenised within the magma ocean depends on the mixing dynamics  
1050 of the magma ocean. The vigor of the convection is essentially governed by  
1051 the viscosity and thickness of the magma ocean. Thus the post-accretionary



1052 homogenisation of the impactor material and its volatile content within the  
1053 magma ocean are, governed by the cooling efficiency of the early Moon. To  
1054 better constrain the fate of the impactor material during a large impact and  
1055 a large accretionary event, we need to carry out 3D models. Such models are  
1056 much more costly in terms of computational time but could strongly constrain  
1057 the influence of the impact angle on the ability of the impact to bury material  
1058 deep within the early Moon and on the fragmentation efficiency. Depending on  
1059 fragmentation of the impactor, the homogenisation could be partial or complete  
1060 (small impactor). A weak homogenisation of the LMO could potentially explain  
1061 the heterogeneous distribution of volatiles in the lunar interior. This aspect is  
1062 beyond the scope of this study and requires dedicated models.

1063

1064 There is no evidence of cratering during the magma ocean period. The im-  
1065 pact craters fade away due to thermal relaxation at high temperature (*Miljković*  
1066 *et al.*, 2021) and the HSE partitioned into the lunar core (*Morbidelli et al.*, 2018).  
1067 Late accretion is considered to be highly stochastic (*Bottke et al.*, 2010) but the  
1068 timing of impacts has a bearing on the potential for burying hydrated mate-  
1069 rial. It is easier to bury small impactors when the crust is thin, and then to  
1070 homogenise them later. As the crust becomes thicker, the critical impactor size  
1071 needed to perforate it increases and the homogenisation of the buried mate-  
1072 rial within a thin magma ocean is more difficult. An approach coupling lunar  
1073 thermal evolution, solidification processes and late accretionary events would  
1074 strongly enhance our understanding of the chemical composition of the lunar  
1075 samples.

1076

1077 Finally, our results show that a significant part of the impactor material is  
1078 molten during the impact for  $v_{imp} \geq 2v_e$  (Figure 9, bottom). As a consequence

1079 its water content may be lost into the lunar atmosphere following the impact. This  
1080 loss may help resolve inconsistent results for considering HSE and H measure-  
1081 ments. Our study focuses on low velocity impacts while high velocity impacts  
1082 are probably more likely at the end of the lunar accretion. Hence we now need  
1083 to understand the evolution of this lunar atmosphere (*Saxena et al.*, 2017) and  
1084 the contribution of the large impacts to its erosion.

1085

## 1086 **8. Conclusion**

1087 We developed hydrocode simulations with iSALE 2D to monitor the ability  
1088 of a large impactor to bury material within a partially molten lunar body. We  
1089 investigated the influence of crustal thickness (from 10 to 80 km), impactor ra-  
1090 dius (from 25 to 200 km) and impact velocity (from  $v_e$  to  $4v_e$ ) on the potential  
1091 contribution of these large impactors on the water budget of the lunar mantle.  
1092 We evaluated the impactor melt fraction assuming chondritic material to de-  
1093 termine the fraction of devolatilised material before penetration into the LMO  
1094 below the solid crust. Our results show that impactors with radii larger than  
1095 25 km are likely to penetrate the crust overlying the lunar magma ocean for  
1096 impact velocities larger than the lunar escape velocity. If the impact velocity  
1097 is higher than  $3v_e$ , complete melting and devolatilisation of the impactor are  
1098 achieved. For these high velocity cases, the volatile content would contribute  
1099 to the volatile budget of the early Moon's atmosphere. However, low impact  
1100 velocities ( $\approx v_e$ ) favour the penetration of the impactor material into the lunar  
1101 mantle without experiencing significant melt and impact dehydration. For these  
1102 cases, at least 90% of the impactors mass may contribute to the volatile budget  
1103 of the lunar mantle.

1104

1105 The mass of late accreted material is commonly calculated using results  
1106 from HSE measurements. We calculated the concentration of water within a  
1107 homogenised magma ocean after an accretionary event consisting of  $2.4 \times 10^{20}$   
1108 kg of chondritic material. We assumed the contribution from two end-member  
1109 types of chondrite : EC (water-poor) and CI (water-rich). The water concen-  
1110 tration in a homogeneous post-accretion mantle would be 1.6 to 10.7 ppm for  
1111 a water-poor case and 168.7 ppm for a water-rich one. Such an accretionary  
1112 event is likely to favour a heterogeneous distribution of volatiles in the lunar  
1113 interior. If post-accretion homogenisation is not achieved, the late delivery of  
1114 volatile elements by impacts could explain the large range of H concentration  
1115 measured in apatites from different lunar rock types (*McCubbin et al., 2023*).

1116

## 1117 **9. Data availability**

1118 The extended data and data analysis from numerical simulations generated  
1119 in this study are accessible via <https://doi.org/10.25519/6PHB-FV69> The  
1120 iSALE-2D input files used to generate simulations are also available and been  
1121 deposited in the same database.

1122

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1135

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Table 1: Parameter values for solid layers

<i>Description</i>	Impactor / Mantle	Crust	References
Equation of State	ANEOS dunite	ANEOS granite	
Initial density	3,315 kg m <sup>-3</sup>	2,630 kg m <sup>-3</sup>	
Poisson ratio	0.25	0.30	<i>Ivanov et al. (2010)</i>
<i>Strength model (ROCK)</i>			
Frictional coefficient (damaged)	0.6	0.8	<i>Ivanov et al. (2010)</i>
Frictional coefficient (undamaged)	1.5	1.1	<i>Ivanov et al. (2010)</i>
Strength at infinite pressure (damage)	3.5 GPa	2 GPa	<i>Ivanov et al. (2010)</i>
Strength at infinite pressure (undamaged)	3.5 GPa	2.5 GPa	<i>Ivanov et al. (2010)</i>
Cohesion (damaged)	0.01 MPa	0.01 MPa	<i>Ivanov et al. (2010)</i>
Cohesion (undamaged)	50 MPa	10 Mpa	<i>Ivanov et al. (2010)</i>
<i>Damage model (Ivanov)</i>			
Ivanov parameter A	10 <sup>-4</sup>	10 <sup>-4</sup>	<i>Ivanov et al. (1997)</i>
Ivanov parameter B	10 <sup>-11</sup>	10 <sup>-11</sup>	<i>Ivanov et al. (1997)</i>
Ivanov parameter C	3×10 <sup>8</sup>	3×10 <sup>8</sup>	<i>Ivanov et al. (1997)</i>

Table 2: Physical parameter values for numerical models

Planet radius	$R$	1,740 km
Magma ocean depth	$z_{LMO}$	300 km
Surface gravity field	$g$	1.561 m s <sup>-2</sup>
Escape velocity	$v_{esc}$	2.33 km s <sup>-1</sup>
Impactor radius	$R_{imp}$	25 - 200 km
Impactor velocity	$v_{imp}$	$1v_{esc} - 4v_{esc}$
Crust thickness	$\delta_{crust}$	10 - 80 km
Surface temperature		293 K
Gradient		0.122-0.0161 K m <sup>-1</sup>
<i>Mantle property (Dunite)</i>		
Initial density	$\rho_m$	3,315 kg m <sup>-3</sup>
<i>Crust property (Granite)</i>		
Initial density	$\rho_c$	2,630 kg m <sup>-3</sup>
<i>Magma ocean (Dunite)</i>		
Initial density	$\rho_{oc}$	3,315 kg m <sup>-3</sup>

Table 3: Resolution parameters for iSALE hydrocode

<i>Domain size</i>	
Horizontal size	2,000 km
Vertical size	10,000 km
Grid spacing	4 km
<i>Cells per projectile radius (CPPR)</i>	
Impactor	25 to 200
Target crust	3 - 25
Target magma ocean	72 - 50
Target mantle	350
Maximum time step ( $dt_{max}$ )	0.05 s